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Chapter 18

Simulation Based Process Design Methods for Maintenance Planning

Joe Butterfield and William McEwan

Abstract The primary objective of this work is to use simulation methods for the development of optimal service support procedures using an integrated Product, Process and Resource (PPR) structure. Mid life fatigue modifications required on a sub-system within an existing aerospace platform ~~that has~~ been used to demonstrate the utility of this approach in supporting the transition of an original equipment manufacturer (OEM) to availability contracting. Other objectives of the work were to show how significant cost drivers can be identified and quantified through the virtual development of work breakdown structures (WBS) for service support processes and to show how the mechanisms used to develop optimal processes, can produce animated instructional materials which enhance organisational learning as processes evolve. This supports the effective communication of methods and work breakdown structures between the technical author and service personnel. Although this work has been completed using an aerospace sub-system as a demonstrator, the outcomes are equally applicable to other platforms which utilise complex engineering systems. The results of this work have system level significance in reducing risk in service provision through the support of value co-creation. The simulation outputs also provide data for higher level cost modelling which in turn, informs the strategy required for supply chain engagement through incentivisation.

18.1 Introduction

The last 10–15 years have seen momentous changes in global socio-economic conditions as well as a significant shift in military strategy as world orders changed after the fall of communism. According to Spreen, this resulted in a drop in

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Fig. 18.1 Evolutionary stages of a learning company (Pedler et al. 1996)

28 demand for military aircraft particularly in Europe during the 1990s and although
29 the emergence of new terrorist related threats prompted a recovery in the demand
30 for defence related equipment in the US after 2001, the probability that the number
31 of new military aircraft programs will recover to cold war levels is exceptionally
32 low. These circumstances have resulted in a contraction in the demand for new
33 product development and an increase in the pressure to keep legacy systems in
34 service for longer. OEMs are therefore, increasingly evolving from product sup-
35 pliers to service providers. Challenging commercial conditions have also resulted
36 in end users re-structuring their own core businesses. A faltering world economy
37 has placed enormous pressures on government finances and despite clear opera-
38 tional needs, defence spending has been highly constrained in recent years. The net
39 result of these circumstances and changing behaviours has been the need for the
40 OEM to *integrate and manage* support service activities *in partnership* with the
41 customer to deliver the availability of a given platform or product. This improves
42 the likelihood of operational sustainability for the OEM through shared com-
43 mercial risks while reducing the cost of ownership for the customer with equiv-
44 alent or improved platform availability.

45 In order to complete the transition to '*availability contracting*', the OEM must
46 move beyond core business activities based on pure manufacture and supply to
47 become involved in the broader product lifecycle. This fundamental shift in
48 commercial practice must take place in partnership with the customer and it rep-
49 represents a significant challenge at all organisational and technical levels within *both*
50 organisations. A successful, sustainable outcome will depend on the extent to
51 which ~~the~~ both parties become a '*Learning Organisation*'. This will require the
52 effective amalgamation of key business elements of the previously separate
53 enterprises and both the OEM and the customer will have to pass through the
54 evolutionary stages proposed by Pedler et al. (1996) as shown in Fig. 18.1, from
55 simply surviving as separate entities, through adaptation to joint sustainability.

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Table 18.1 Characteristics of the Learning Company

Learning approach to ← strategy	Pilots and experiments used to inform direction
Participative policy ← making	All stakeholders contribute to policy making
Informing	Use of I.T. to automate and distribute information thereby empowering ← operatives to act on their own initiative
Formative accounting ← and control	One aspect of informing opening up budgeting, reporting and ← accounting to broader enterprise
Formative accounting ← and control	One aspect of informing opening up budgeting, reporting and ← accounting to broader enterprise
Internal exchange	All internal units and departments are effectively customers and ← suppliers in a supply chain leading to the customer. Contracting ← with and learning from one another
Reward flexibility	Greater participation creates the need for flexible and creative ← monetary/non- monetary rewards
Enabling structures	Roles, departments, organisational charts, procedures and processes ← can be viewed as structures that can be used to meet job, user or ← innovation requirements
Boundary (interface) controls	Used for 'environmental scanning' by those who have contact with ← external users, customers, suppliers, clients, business partners etc. ← Processes used for data handling passing into (and out) of the ← enterprise
Inter-company learning	Through joint ventures and other learning alliances, organisations ← learn from other companies. This is facilitated by joint meetings for ← mutually beneficial information exchange
Learning climate	Management facilitating experimentation and learning from ← experience through questioning, feedback and support. This is ← subsequently exported to the broader commercial and technical ← activity and any business partner

justify.?

56 A learning organisation is one that can change from within itself. Easterby-
57 Smith (1999) states that implicitly, its underlying principle is an adaptive one
58 where adaptation to environmental conditions or pressures leads to organisational
59 learning and changes. Learning and changing faster than others can help to ensure
60 the maintenance of any competitive edge or at the very least, increase the orga-
61 nisation's chances of survival. According to Pedler et al. (1996) the characteristics
62 of the learning company are listed in Table 18.1.

63 This work has come about as a result of the recognition that evolution or
64 transition to availability contracting will become a significant part of the opera-
65 tional activities of BAE Systems moving forward. BAE Systems is already
66 evolving into a 'learning organisation' and the characteristics listed in Table 18.1
67 are all engendered in the EPSRC funded program entitled Support Service Solu-
68 tions: Strategy and Transition (S4T). The material presented in this chapter was
69 carried out as part of Work Package 3 within this program. It presents a strategy
70 for the virtual simulation of maintenance planning processes and their execution
71 for enhanced service provision. The main outputs required were WBS, the iden-
72 tification and quantification of sub-system level cost drivers (for subsequent use in

Fig. 18.2 Position of sub-system simulation activity within work package 3

73 system level cost models) and the provision of digital instructional materials.
74 Digital manufacturing tools were used to deliver these requirements. The data flow
75 diagram showing how simulation fits with the other research streams within Work
76 Package 3 can be seen in Fig. 18.2.

77 Butterfield et al. (2007) have stated that the need to minimise the risk of
78 commercial failure by developing leaner products has lead to the development of
79 manufacturing simulation software systems which use simultaneous or concurrent
80 engineering concepts for the design and engineering of new products across a
81 range of industries. The way in which any complex assembly is constructed or
82 de-constructed for maintenance, repair or overhaul (MRO) is an important con-
83 sideration if a whole life approach is taken to product design and process devel-
84 opment. Traditionally process design has not been supported by predictive
85 technologies to the same degree as other areas of product design and development.
86 To deal with this problem concurrent engineering concepts are now available in
87 computer aided design (CAD) and computer aided engineering (CAE) tools and
88 they are used to identify optimal *process* designs. Although to date, these tools
89 have been used in a *product* design and manufacturing context, they are equally
90 applicable to *process* design for the maintenance repair and overhaul (MRO)
91 activities that are at the core of availability contracting. The simulation techniques
92 that these systems offer allow process planners to define, validate, manage and
93 deliver fully optimised manufacturing or MRO process specifications.

94 Although software vendors can provide 'off the shelf' solutions for the
95 analysis of typical manufacturing processes, very often OEMs have company
96 specific activities and requirements which do not fit into standard software
97 versions. The CATIA¹ (Computer Aided Three-dimensional Interactive

¹ CATIA is a 3D CAD modelling environment used for the design and specification of engineering components.

Fig. 18.3 Pintle frame
location within the front
section of Panavia Tornado

98 Application) and DELMIA² (Digital Enterprise Lean Manufacturing Interactive
99 Application) software modules used for this work were enhanced and customised
100 where required, to suit the specific requirements of service support process
101 development. An exemplar is presented which provides an evaluation of simu-
102 lation methods on a 'proof of concept' basis. The exemplar is based on a mid
103 life fatigue modification required on a sub-system within the twin-engined
104 Panavia Tornado combat aircraft. The work shows how digital methods, origi-
105 nally developed for purely manufacturing applications, can be applied to a low
106 volume/high complexity task based on the replacement of a pintle frame section.
107 The pintle is a significant piece of internal structure in the front end of the
108 aircraft, see Fig. 18.3. The application in this instance is based on the simulation
109 of the removal and replacement of key structural components in order to
110 demonstrate how the entire process can be designed and captured in a virtual
111 environment. This, in turn, demonstrates how the methods used can be applied to
112 the service of elements within a complex asset which were never designed to be
113 replaced. This is an important example of the increasing complexity of defence-
114 based service provision.

115 The work focuses on the development and validation of an optimal WBS and
116 the main outputs are the process definition including task completion times
117 (to facilitate cost prediction), the development of digital animations for process
118 design purposes as well as instructional delivery. Butterfield et al. (2007) have
119 demonstrated the utility of the method in a manufacturing context for improving
120 operator learning through the use of animated work instructions. Butterfield et al.
121 (2008) have shown improved management learning using concurrent digital
122 working practices and Jin et al. (2009) for simulating manufacturing assembly times.

123 The work detailed in this chapter uses a CAD-based collaborative environment
124 to develop this approach to bring simulation and optimisation strategies in line
125 with the planning required for aerospace-based availability contracting at BAE
126 Systems. The structure uses, defines and retains process data for a single sub-
127 system in this case, but the hierarchical nature of the structure means that complete
128 systems can be modeled. The end result is an integrated, collaborative environ-
129 ment containing the WBS which is developed from the manufacturing bill of

² DELMIA is an integrated, 3D CAD based environment used for engineering process design and optimisation.

Fig. 18.4 Transformational bridge between individual and organisational learning, Probst

CAPITAL 'P'
for Probst.

materials (MBOM). It defines the processes associated with service support activities and includes time and cost data for product assembly and dis-assembly as well as the animated process definitions. According to Probst and Buchel (1997) this then places the digital tools used for this work, in the role of a 'bridge' between the behaviours associated with the types of individual learning discussed by Butterfield et al. (2008) (managerial, operator etc.) and those required for the system level organisational learning which will aid the transition to availability contracting, see Fig. 18.4.

18.2 Sub-System Level Simulation

18.2.1 Background

The recent extension to the service life for the Tornado has given rise to the Mid Life Fatigue Programme (MLFP) within BAE Systems. A physical test programme has identified several critical airframe components which are approaching the end of their fatigue life. The original service expectation on the aircraft meant that these components were designed to operate up to the temporal point now reached by the aircraft. The extension to service means that these components must now be replaced.

The original design constraints and demands on these components did not include repair or replacement as they were expected to perform within design tolerances for the specific life span originally dictated. Many of these components are embedded deep within the airframe superstructure. Issues of access and method of change remain the dominant feature of the MLFP programme. Limitations of available data, as well as test case aircraft to assess concepts represents a strong case for digital simulation of MLFP activities in order to arrive at optimum, cost-effective solutions.

Fig. 18.5 Pintle frame location

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For the purposes of this work, Queen's University Belfast (QUB), in conjunction with BAE Systems Warton, have digitally simulated the replacement process of a major airframe structural component, the pintle frame, with a view to establish an optimum solution in both technique and cost. Utilising an integrated PPR structure, it was possible to model the relevant airframe components, associated fixtures and spatial constraints using DELMIA V5. The working methodology through the BOM within the DPM (Digital Process for Manufacture) environment and the associated work flow of the physical methodology. One specific proposal was then animated and ergonomics and clash detection utilised to establish the WBS.

18.2.2 Test Case 'The Sub-System'

Figure 18.5, shows the major portion of the airframe investigated. The panel to the rear of the seatwell box assembly, shown in exploded format, is the pintle frame.

Fig. 18.6 Simulation of work space environment surrounding pintle frame

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18.2.2.1 Establish Maintenance Activities as Part of Standard BAE Systems Protocol

Generic fitting activities are well established under the normal operating routines at BAE Systems. Pintle frame replacement constituted a significant challenge even with experienced fitting operatives. To gain access to the panel prior to replacement, it was first necessary to dis-assemble and remove the undercarriage. With access to the seatwell established, the hardware bolted to the interior surfaces of the surrounding structures was also removed and/or loosened from position to gain access to the pintle frame and create space within the seatwell, for the technician's working convenience.

From this point onwards, all activities are modelled within the digital environment. All movements, associated tooling and fixtures were used as input of time dependant variables representing the sub-system level activities contained in the pintle replacement work package.

input as.

18.2.2.2 Assembly Model of Environment

Beginning with the major structural components, provided by BAE Systems in CATIA V4 format, all CAD data was translated into the V5 environment. A set of idealised shear walls were added to simulate the surrounding structural elements (see Fig. 18.6) for which only 2D data in the form of lofted drawings was available. This was the standard method for providing engineering data for a given part at the time when the Tornado was designed.

With the interior working volume established, the exterior surfaces were modelled to accurately reproduce the working environment. Again a simple idealised surface model of the exterior of the forward fuselage skin was used to provide relevant information pertaining to the movement of technicians, tooling

Fig. 18.7 Human analysis simulation in pintle frame area

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194 and fixtures around the work area. Information related to the distance between the
195 workshop floor and the fuselage model was required to establish the limitations on
196 technician movement around the work area.

18.2.2.3 Human Considerations

198 The shear wall surfaces were idealised to provide the physical boundaries or
199 ergonomic constraints for the human analysis model. Figure 18.7 shows how a
200 mannequin was used make an ergonomic assessment of the work area in front of
201 the pintle frame. Here the mannequin can be seen in a position of disadvantageous
202 medial rotation. This resulted from the lateral limitations of the work space in
203 conjunction with the forced positioning with respect to the work piece height from
204 the hanger floor. This example is an ideal illustration of how the effect of physical
205 limitations can be analysed to express the optimum solution ~~to~~ operator motion. In
206 this case the analysis of posture indicates a limitation on lateral and transverse
207 application of force available to the operator, and hence a clear design choice to
208 focus on longitudinal motions to execute the task. In response, the associated
209 tooling designed ~~to~~ meet this particular objective was designed with longitudinal
210 motion as the key ergonomic requirement.

18.2.2.4 Identification of Parts and Fasteners

212 Fastener type and count were recorded based on the original manufacturing BOM.
213 It must be noted here that CAD definitions of the fasteners does not exist and is not
214 common practice in aerospace design. Fastener type, number and associated time
215 for removal and replacement, however was defined for the purpose of time gen-
216 eration and their locations were determined from the 2D data provided by BAE
217 Systems. To improve the efficiency of this task, a Visual Basic script was also
218 devised based on an automated search which recovered hole sizes, quantities and
219 depths directly from the feature tree in the CAD model see Fig. 18.8. Visual Basic

Fig. 18.8 Automated extraction of part data from feature tree in CATIA

is a programming language that enables the development of graphical user interface applications as well as access to databases using Data Access Objects, Remote Data Objects, or ActiveX Data Objects. In simple terms, this was a computer program designed to extract hole properties (sizes, quantities etc.) from the CAD model. By associating fastener types with given hole sizes, fastener counts could be established. The hole size and depth was equated directly with drilling time.

18.2.2.5 Develop Simulation for Pintle Replacement

Initial product assessments showed that the major dis-assembly of the airframe that was required to free the pintle frame for replacement was neither practicable nor cost effective. From this, the decision to re-work the pintle in situ was made. From this position, the first design constraints were established for any proposed engineering solutions.

A preliminary trade-off study was performed to establish the optimum method for the pintle replacement from a high level engineering perspective. The three solutions considered were:

- *Full Automation*—Computer controlled robotic machine installations
- *Semi Automation*—Hand held machine tools and guide templates
- *Manual*—The use of manually guided hand tools only

Each of the potential methods was rated according to six different metrics:

- *Time*—Required to perform tasks including set up, measurement or metrology and inspection; all value added activities
- *Re-use*—A measure of the potential utilisation of any hardware for future tasks not associated with the pintle replacement

Fig. 18.9 Results of trade off study to establish high level methodology

- *Repeatability*—The recurrent holding of tolerances in all worked parts
- *Accuracy*—The capacity to meet geometric requirements with regard to the position of all worked surfaces with respect to existing geometry
- *Ergonomics*—Ease of operator movement (Includes measure of fatigue)
- *Capital outlay*—Cost of any associated hardware or consumables

The graphs shown in Fig. 18.9 illustrate the overall result of this decision making process. Graph (a) shows how each of the three methodologies performed with regard to the six metrics listed previously, as a mark out of a possible

maximum of five where: 1 = worst/low rating, 5 = best/high rating. Graph (b) in Fig. 18.9 shows the weighted average of these values as a non-dimensional expression between 0 and 1. This graph shows that, for the conditions used in this trade study, the semi-automatic methodology had the best combination of performance properties.

Having established the benefits of a semi-automated solution, a secondary set of design criteria were established. The need to escape the uncertainty in both repeatability and accuracy with the fully manual methodology, as well as the intrinsic high temporal cost, predicated a maximum utilisation of suitable machine tools. The limitations on such tooling, while including capital cost, were driven primarily by physical size. A survey of available hardware, performed for the trade off study, identified a number of miniaturised milling machines capable of carrying the appropriate tool bits and performing with the necessary power to cut the maximum skin thickness identified from the pintle geometry.

18.2.2.6 Fixtures and Tooling

The crux of the engineering problem presented here involves the fact that the replacement pintle pieces are newly manufactured. These have to be fitted to pre-existing, well used airframes of variable, and most importantly un-quantified, dimensional stability. With this in mind, a decision was made to use a machining pattern or fixture that would facilitate accurate and repeatable removal of the redundant pintle material.

With the idealised machine tool modelled in the CAD environment, to provide the required dimensions for the interface with the new piece, it was possible to test the positions and motions of the tool to establish the optimum path. This was carried out as follows:

- *Tool path*—derived from the final geometry of the portion of the pintle which was to remain in situ, and the replacement piece to be introduced and fitted. The minimum number of movements and set ups required by the operator in execution of cutting. Clash detection with all surrounding components with respect to the mill dimensions
- *Tool choice, cutting speed and feed rate*—again derived from the final geometry, variation in material thickness, minimum tool movement during operation and best possible proximity of the mill to the work piece without clash

Choice of tooling implemented in the final design solution was established iteratively through the simulated environment including design, assembly and human analysis methods. For example, the mini-mill chosen for the machining operations was one of a small number of machines initially selected for its size characteristics. All primary tool choices had to be small enough to fit within the confined space of the pilot seatwell with the operator. From within the design environment, the tool path was established for the removal of material from the original pintle. From the assembly environment, it became apparent that the

proximity of the shear webs and pintle bush lugs to the tool path negated the use of some of the primary tooling choices due to a physical clash between the tools and the surrounding structure. From the human analysis environment, it could be seen that the physical limitations on the operator posture resulted in a severe reduction in the operator's capacity to apply any meaningful degree of lateral force when handling the tool. From this an engineering decision to select a tool which could be handled in a predominantly longitudinal manner was made.

One major problem associated with the overall task was the establishment of reliable datums on which to base all hardware. The replacement sections of the pintle frame will be manufactured off site using a fixed set of dimensional constraints, but the surrounding airframe components and their position with respect to one another cannot be assumed to be as per original design or replicant from one aircraft to another. With these conditions in mind, it was decided to select the main undercarriage bearing positions on the longerons as the most suitable datum. Since the longerons are major structural components responsible for the positioning of much of the rest of the airframe, it can be assumed that they are prone to the least digression from original specification, both in terms of intrinsic dimensions, and also with regard to position in respect to one another. By extension it is logical to assume that the digression of the pintle frame position with respect to the longerons is also limited. Finally, since the pintle frame shares the same bearing locations as the longerons it is sensible to fix the position of the new pintle section from these points and base the cutting geometry of the original pintle with respect to the insert. This ensures that when the new section is inserted and located correctly with respect to the bearing locations, the machined surfaces on the cut out of the original pintle are aligned correctly with the mating surfaces of the insert, regardless of any distortion present in the original pintle.

The design solution arrived at consists of a flat guide plate which locates on the bearing positions and alters between a horizontal position for the early cutting stages and vertical orientation for the latter stages. This plate has guide paths on its surface which correspond to pins on the mill allowing it to follow set paths which mimic the required cut out profile on the pintle frame. The translation between horizontal and vertical positions is to allow the change in orientation necessary to make the set of cuts that exist in different planes. This plate, like the mill, is available for repeated use on each aircraft without the need for any measurement or alteration. This particular feature represents a large time saving in the overall time required to perform the task. In addition to this plate, a drilling template for the positioning of holes in the outer webs of the pintle is also utilised.

Major structural components, idealised bounding surfaces and where data is available, the sub-components fastened to the pintle frame were assembled within the CAD environment. Parts, fasteners and the positioning of all relevant components were fixed. The BOM, held within the PPR hub, was then available for interrogation by the user. The final choice in milling machine was the *Ricci Corp Mini-mill*.

For the given material properties of the components and the maximum material thickness to be encountered, engineering rules were applied and a size up regime

Fig. 18.10 Pintle frame with web material still in place

for the drilling operations was established as 2 mm to 3.5 mm to 5 mm. A long taper flute drill is best suited to this material, however, the time deriving software for drilling operations that was used in this project, SEER Design for Manufacture,³ does not support such bits, hence, normal twist drills were used in this model. Since drilling templates were used where new holes were to be established without a pre-existing guide hole in a mating component, there was no requirement for spot drilling to ensure accurate hole placement. This represents a classic engineering decision within this model, to create a process which requires a minimum of specialised skill on the part of the operator, and hence mitigating against problems associated with the availability of highly skilled operators.

As mentioned previously, drilling operation times were established via SEER where the times associated for the drilling of any specific hole was established as a factor of drill size, hole depth, material properties and rotational speed of the drill. In a similar manner, the cutting speed of the mill was established through SEER also. The first milling operation involves the removal of the major portion of extraneous material from the pre-existing pintle. A 2 mm diameter plunge and cut bit of high speed steel with 4 flutes was chosen for this task. SEER has the capacity to model milling operations according to tool specifications (as listed above), the manner of cutting and the length and depth of the cutting path, or alternatively the volume of material to be removed. As the depth of the material, through any given cutting path was not consistent, each cut was modelled as a separate operation, the number of which was equal to the number of step changes in material depth through the entire cutting path. This provided optimal accuracy in the output time from the SEER model.

With the first cut complete and the waste material removed, the mill was fitted with a 10 mm diameter edge cutting bit of high speed steel with 8 flutes. With this tooling the internal edge of the pintle was finished to final dimensions. The final milling operation involves the removal of web material from the internal face of the pintle. The web material is highlighted in Fig. 18.10.

A 28 mm diameter face and edge cut tool of high speed steel and 16 flutes was selected. The final face profile is shown in Fig. 18.11. A three stage cutting plan was followed which avoided the possibility of machining in-accuracies due to removing too much material in one run.

³ SEER is a registered trade mark for cost management software marketed by Galorath inc.

Fig. 18.11 Pintle frame in final state

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Fig. 18.12 Placing of mill and tool lateral cuts

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Fig. 18.13 Introduction of new pintle part

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Each discrete drilling operation was modelled within DPM where the tooling can be seen moving around the work piece as per the process order. This in turn is also seen in the output work instructions.

18.2.3 Process Definition

The dis-assembly activities and all associated data were established using the simulation methods detailed above. A preliminary discussion with BAE Systems led to the following broad work stream order: Fastener strip, disassembly and removal of sub-components, clean down of exposed pintle frame, introduction of fixture(s) for cutting, machining and dressing, cut and dressing process for pintle frame, removal of now extraneous material from pintle frame, clean down, assembly of new pintle sub-part, fastening, re-assembly of sub-components, fastening and inspection. Each of these activities was broken down into sub-activities with different requirements in tooling, labour and hence time. The sub-activities relating to the work focussed directly on the pintle frame are derived directly from the simulation as optimised solutions.

Figures 18.12 and 18.13, show 2 stages of the process required to replace the pintle frame.

Fig. 18.14 Inputs, process and outputs resulting from sub-system level simulation

Each discrete component movement was animated within the CAD environment as they were removed or replaced. Clash detection studies were carried out to establish optimum pathways for each component movement. In conjunction, mannequins were utilised to perform an ergonomic assessment of the simulated activities.

With a working order established to the physical activities, each was recorded within Delmia Process Engineer (DPE)⁴ as discrete actions with their associated time and tooling requirement.

18.2.4 Simulation Outputs

A number of outputs were derived directly from the digital simulations. These were: Optimised work breakdown structure, identification of cost drivers, delivery of representative times for activities and the generation of instructional materials.

Figure 18.14 shows the overall process structure for the use of simulation methods to yield these outputs. Section 2.3 details the WBS which was defined in the CAD environment and subsequently recorded in DPE. The WBS was supplemented with process times and part counts.

The second output from this work package component was the identification of cost drivers. The drivers arising from the parameters within the sub-system simulation were found to include: Replacement parts, tooling, fixturing, fasteners and operator time (for dis-assembly/assembly and inspection/quality control).

The timing data generated from the simulations is in the form of the 'hands on' time associated with specific activities. When this is used for process costing at a

⁴ DPE is the process modelling arm of the Dassault Systemes suit where discrete processes can be modelled, in order, and with associated attributes such as time.

410 system level, these times must be supplemented with data associated with labour
411 type and source, i.e., BAE Systems staff or customer personnel. A utilisation factor
412 must also be applied to allow for 'non hands on' activities which are an integral
413 part of an operator's working day. Consideration of learning curve impact is also
414 critically dependant on labour type. Both utilisation and learning curve allowances
415 will vary depending on the labour source and should be an important part of the
416 system level costing strategy.

417 The final deliverable is represented by the animated work instructions. Clash
418 detection, and ergonomic considerations, generated the most time conservative
419 solution to the task of pintle frame repair. From the same digital simulation,
420 instructional materials were generated to assist the execution of the pintle
421 replacement activities. Butterfield et al. (2008) have shown that the use of ani-
422 mated instructional materials can reduce the learning curve resulting in lower
423 manufacturing costs. It is expected that a similar result can be achieved with regard
424 to maintenance and repair activities if operator learning is driven by animated
425 work instructions. Here the process order and the specific details of each process
426 have been shown as virtual movement of the components complete with fixtures
427 and tools, as ~~his~~ would occur in the real world scenario. Accompanying these
428 animations are written instructions which appear as dialogue boxes at the points
429 most appropriate to the specific processes. These instructions contain the details
430 that are not immediately apparent from the animation. This includes such details as
431 specific tool size or positioning of fixturing.

432 18.3 Discussion

433 The transition from a conventional supplier/customer relationship to an availability
434 contract arrangement, constitutes a significant challenge for both parties at all
435 levels within the enterprise—commercial, technical and human. According to
436 Butterfield et al. (2009), the nature of the commercial relationship changes as
437 previously separate business entities merge and self interests are replaced by the
438 need for optimal 'co-performance'. Operational sustainability remains a challenge
439 for any business entity under current political and economic conditions. Compa-
440 nies must become learning organisations which can react efficiently to changing
441 market needs to retain any operational advantage or in extreme cases, simply to
442 survive. The fact that BAE Systems have recognised the need for and initiated the
443 S4T program means that it is evolving as a learning organisation. The charac-
444 teristics listed in Table 18.1 are engendered in the S4T program as a whole and the
445 simulation methods used for this work can be linked directly or indirectly to most
446 of the same characteristics.

447 The aerospace exemplar presented here shows how simulation and the princi-
448 ples of digital manufacturing can be used to design service solutions for complex
449 platform elements which were never intended for replacement. The functionality
450 of the simulation framework allows complete system definition through the

451 hierarchical summation of process definitions and parameters within individual
452 sub-systems. The development, optimisation and retention of processes knowledge
453 in the virtual environment, increases the likelihood of a positive outcome for both
454 the OEM and the customer by providing a framework by which the attributes of
455 value co-creation discussed by Ng and Nudurupati (2010) and Ng et al. (2010a),
456 can be facilitated using a controlled, measureable methodology. This enables the
457 transformation of process structures (*information*), product definitions (*materials*
458 *and equipment*) and people (*instructional materials*) in line with what will be
459 required for the provision of platform availability. The simulations deliver process
460 details from an engineering perspective, which include WBS for maintenance
461 operations, delivery of representative times for 'hands on' activities (which can be
462 used for resource optimisation and cost analysis), identification of cost drivers such
463 as labour, tooling, fixturing and inventory and the generation of instructional
464 materials. The outcomes can be used in turn, for system level cost modelling. The
465 approach to the delivery of instructional materials in this case has been to use
466 animated maintenance processes. The ability to carry out significant modifications
467 in small batches is an important factor in service provision. This emphasises the
468 need for an efficient means of transferring knowledge and instructing operators
469 who may not be under the direct control of the OEM. In a service context they may
470 have to carry out multiple tasks on a given system or across several platforms. The
471 availability of animated instructional materials facilitates the operational agility
472 and breadth of knowledge that an operator must have to deliver these efficiently.

473 The digital methods presented here are best suited for complex 'hard' engi-
474 neering problems such as the pintle frame replacement. For other test and
475 inspection activities the added value of this approach is questionable but the digital
476 systems used for this work can still play a part in developing and delivering
477 process definitions. Test and inspection activities can still be supported using a
478 digital approach but the instructions would be based on static CAD images.

479 Digital manufacturing provides an integrated framework for the generation,
480 management and delivery of engineering process definitions. Within a single
481 enterprise, this facilitates concurrent activities between the engineering disciplines.
482 The transparent and collaborative environment can be enhanced to include
483 maintenance activities, as the process node in the manufacturing hub is extended to
484 include the broader product lifecycle. The definition and optimisation of mainte-
485 nance processes in a virtual environment using virtual assets reduces the uncer-
486 tainty associated with availability contracting. Simulated process scenarios allow
487 the process designer and implementer to reduce or eliminate exposure to risk and
488 uncertainty by generating realistic data for important cost components.

489 This work has shown how digital methods can be used in support of the transition
490 to availability based contracting. The benefits of using process simulation methods in
491 support of this business model has been demonstrated on a proof of concept basis.
492 The next step would be the implementation of the methods developed here starting
493 with a pilot program covering a broader platform application working from sub-
494 system to system level simulations. In order to exploit the utility of the methods
495 presented here the question of how the simulation platform is managed has to be

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addressed. Process data related to MRO activities is generated at source by the platform manufacturer. When held as a central data repository within the PPR hub, process data can be accessed by individuals and disciplines across the enterprise. Levels of access and user privileges can be controlled by setting the relevant permissions. With current software versions, this can be achieved across existing enterprise wide networks but access can be a problem for outside parties due to security set ups etc. Data should also be available to both OEM and the customer and this has been recognised as an issue. Future versions of the software will be web mounted making access between enterprises possible.

18.4 Conclusions

A sub-system level exemplar based on the partial replacement of an aerospace structural panel has shown how simulation can deliver virtual process definitions for service support activities including work breakdown structure, key cost drivers and instructional materials. The utility of manufacturing simulation techniques in maintenance process design has been demonstrated. This requires the extension of a hierarchical PPR environment into the extended platform lifecycle where the assembly and dis-assembly processes associated with MRO activities can be modelled. The resulting process structure enables the determination of process sequences, labour hours and part counts. Instructional materials are developed concurrently with process design thereby eliminating the need for separate instructional authoring. This has been demonstrated at a sub-system level and the outcomes can feed into system level cost models. The PPR structure facilitates the process of knowledge generation, capture and retention thereby facilitating the transition of an OEM to availability contracting as a learning organisation.

The outcomes integrated with the other WP3 themes with digital methods supporting the Attributes of Value Co-Creation ~~was~~ developed by Ng et al. (2010a). The identification and quantification of sub-system cost drivers provided inputs for the cost modelling activity carried out by Roy et al. (2009), which in turn, informed the incentivisation work carried out by the University of Bath. Outside WP3, the QUB simulation outcomes provide the 'What?' and the 'How?' in terms of service support process definition and this information can be integrated with the University of Salford, WP4 activities related to the question of 'When?' in availability contracting.

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18.5 Chapter Summary Questions

Having demonstrated the utility of digital methods for service provision on complex assets, any decision to move forward with this technology should be based on the outcomes presented here as well as the following:

- 533 1. To what extent will existing customer assets be expected to perform beyond
534 their original design life?
- 535 2. How compatible are original platform designs with the types of service required
536 to extend their design life?
- 537 3. If provision for 1 and 2 above is required how will this be addressed in
538 emerging services?

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